

PART "B" VARIABLE LOAD

The weight and lever arms of the variable load are shown below. The variable load depends upon the equipment carried for the particular role

ITEM	Weight lb.	Lever Arm inches.	Moment lbin/100
..... Crew member @ 165 lb per person **		85.5	
Row 1 Passenger Seat ()		85.5	
Row 2 Passenger Seats ()		119.1	
Row 3 Passenger Seats ()		157.6	
Navigation Bag ()		178.7	

** The actual weight and moment of the pilot may be used provided that the total weight and moment of variable load is adjusted accordingly.

PART "C" - LOADING INFORMATION (Disposable Load)

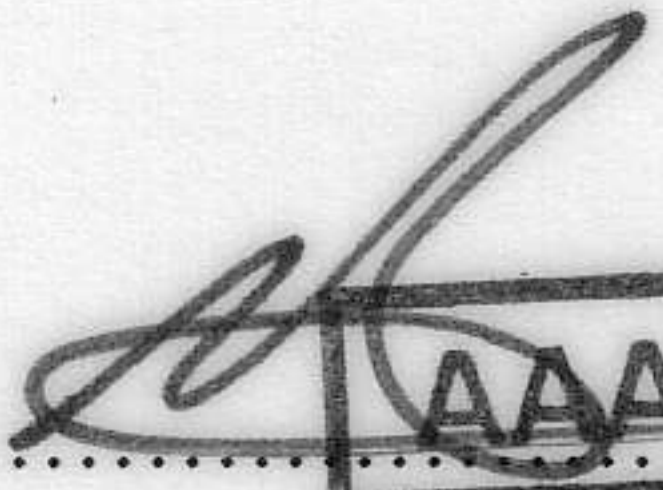

The total moment change when the landing gear is retracted in lbs ins/100 is:

ITEM	Weight lb.	Lever Arm inches.	Moment lbin/100
Fuel in tanks - MAIN * 48 gallons maximum		94.0	
Engine Oil *	Included in	basic weight	
Baggage (Fwd) 100 lbs maximum		42.0	
Baggage (Aft) 100 lbs maximum		178.7	
Passengers in - Row 1 seat		85.5	
Passengers in - Row 2 seats		119.1	
Passengers in - Row 3 seats		157.6	

* Fuel density 7.2 lb/gal and Oil density 9.1 lb/gals.

NOTE: The total loaded weight of the aircraft is the sum of the operating weight of the disposable load.

This schedule was prepared on **30 January 1997** and supersedes all previous issues.

Signed:   Chief Engineer for Anglo American Airmotive Ltd